UNCLASSIFIED

AD NUMBER AD803042 LIMITATION CHANGES TO: Approved for public release; distribution is unlimited. FROM: Distribution authorized to U.S. Gov't. agencies and their contractors; Administrative/Operational Use; NOV 1966. Other requests shall be referred to Air Force Machinability Data Center, Cincinnati, OH. AUTHORITY USAFML ltr 21 Jun 1967

AIR
FORCE
MACHINABILITY
DATA
CENTER

DEC 5 1966



NOTICES

When U. S. Government drawings, specifications, or other data are used for any purpose other than a definitely related Government procurement operation, the Government thereby incurs no responsibility for any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise, as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may be related thereto in any way.

Qualified requesters may obtain copies of this report from DDC, Document Service Center, Cameron Station, Alexandria, Virginia 22314. Orders will be expedited if placed through the Librarian or other person designated to request documents from DDC.

DDC Release to CFSTI not authorized. The distribution of this report is limited because it contains technology identifiable with items on the strategic embargo lists.

Reproduction in whole or in part is prohibited except with the permission of the Manufacturing Technology Division. However, DDC is authorized to reproduce the document for "U.S. Governmental Purposes".

Do not return this copy unless return is required by security considerations, contractual obligations, or notice on a specific document.

BLANK PAGE

AIR FORCE MACHINABILITY DATA CENTER

Cincinnati, Ohio 45209

MACHINING DATA FOR NUMERICAL CONTROL

REAMING - AFMDC 66-1.7

(SUPPLIED IN RESPONSE TO YOUR REQUEST CHECKED / ON USER FILE ANNOUNCEMENT NO. 1)

COST OF ADDITIONAL COPIES

0F

MACHINING DATA FOR NUMERICAL CONTROL

REAMING - AFMDC 86-1.7

Number of Copies	Price per Copy
1 - 9	\$1.00
10 - 49	.90
50 - 99	.80
100 or Over	.70

Please direct your Purchase Order to:

Air Force Machinability Data Center 3980 Rosslyn Drive

Cincinnati, Ohio 45209 Attn: Mrs. Mary Jane Finn Supervisor, User File Michael Field Clarence L. Mehl John F. Kahles

NOVEMBER 1966

Advanced Fabrication Techniques Branch
Manufacturing Technology Division
Air Force Matarials Laboratory
Research and Technology Division
Air Force Systems Command
United States Air Force
Wright-Patterson Air Force Base, Ohio

This document is subject to special export control and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Manufacturing Technology Division.

MACHINABILITY DATA PRODUCTS AND REPORTS

AIR FORCE MACHINABILITY DATA CENTER (AFMDC)

COPIES OF THESE REPORTS MAY BE OBTAINED FROM AFMOC UNTIL THE SUPPLY IS EXHAUSTED. ONE COPY OF EACH REPORT IS AVAILABLE, WITHOUT CHARGE, TO THE AEROSPACE INDUSTRY, DEPARTMENT OF DEFENSE (INCLUDING ALL OF THE MILITARY SERVICES AND THEIR CONTRACTORS), AND OTHER GOVERNMENT AGENCIES, TECHNICAL INSTITUTIONS, AND NORMILITARY INDUSTRIES IN A POSITION TO ASSIST THE DEFENSE EFFORT. PRICE LISTS FOR ADDITIONAL COPIES ARE PROVIDED WITH EACH REPORT.

QUALIFIED REQUESTERS MAY ALSO ORDER COPIES OF THESE REPORTS FROM THE DEFENSE DOCUMENTATION CENTER (DDC), CAMERON STATION, ALEXANDRIA, VIRGINIA 22314.

REPORT NO.	TITLE
AFMDC 65-1	MACHINING DATA FOR TITANIUM ALLOYS, AUGUST 1965, AD-623 588
AFMDC 65-2	FIRST ANNUAL REPORT OF THE AIR FORCE MACHINABILITY DATA CENTER,
	FEBRUARY 1966, AD-482 278
AFMDC 66-1.1	MACHINING DATA FOR NUMERICAL CONTROL - TURNING, JUNE 1966, AD-483 994
AFMDC 66-1.2	MACHINING DATA FOR NUMERICAL CONTROL - FACE MILLING, AUGUST 1966, AD-487 156
AFMDC 66-1.3	MACHINING DATA FOR NUMERICAL CONTROL - DRILLING, AUGUST 1966, AD-488 018
AFMDC 66-1.4	MACHINING DATA FOR NUMERICAL CONTROL - PERIPHERAL END MILLING, SEPTEMBER 1966
AFMDC 66-1.5	MACHINING DATA FOR NUMERICAL CONTROL - END MILL SLOTTING, SEPTEMBER 1966
AFMDC 66-1.6	MACHINING DATA FOR NUMERICAL CONTROL - TAPPING, NOVEMBER 1966
AFMDC 66-1.7	MACHINING DATA FOR NUMERICAL CONTROL - REAMING, NOVEMBER 1966
AFMDC 66-2	GRINDING RATIOS FOR AEROSPACE ALLOYS, JUNE 1966, AD-483 995
AFMDC 66-3	MACHINING DATA FOR BERYLLIUM METAL, JUNE 1966, AD-485 297
DATA PRODUCTS I	N PREPARATION:
AFMDC 66-1	MACHINING DATA FOR NUMERICAL CONTROL (Collection of 66-1.1 through 66-1.7 in one volume)

TABLE OF CONTENTS

INTRODUCTION	٧
REAMING - INDEX TO DATA TABLES	224
REAMING - DATA TABLES	225-233
APPEND I X	
CUTTING FLUIDS USED IN TESTS	A-2
R = RECOMMENDED CUTTING SPEED	A-2
CARBIDE GRADE CHART	A-3
IDENTIFICATION AND TYPE CLASSIFICATION OF HIGH SPEED TOOL STEELS	A-4
CAST ALLOY TOOL MATERIALS	A-5
TOOL GEOMETRY - REAMING NOMENCLATURE	A-6
ROCKWELL-BRINELL - ULTIMATE TENSILE STRENGTH HARDNESS CONVERSION CHART	A-7
DEFINITIONS - METALLURGICAL AND HEAT TREATING TERMS	A - 8
DESCRIPTION OF AFMDC	A-17
OTHER SOURCES OF MACHINING DATA	A-18

BLANK PAGE

INTRODUCTION

5年,中国的第三人称单数

This report presents an extensive set of machining data selected from six USAF Machinability Reports. Data are tabulated and arranged in formats including all pertinent machining variables. The relationship between tool life and cutting speed is expressed in at least three sets of data, thereby making it possible to optimize for maximum production or minimum cost. While these data are expected to be of considerable assistance in providing data for numerical control applications, they are also of great value in any type of machining situation involving the materials for which machining data are presented. Specifically this report, No. 66-1.7, the final of this series, pertains to reaming. The preceding reports in this series were: turning, No. 66-1.1; face milling, No. 66-1.2; drilling, No. 66-1.3; peripheral end milling, No. 66-1.4; end mill slotting, No. 66-1.5; and tapping, No. 66-1.6.

Each report is distributed upon completion, and all reports for the various machining operations will be collected in a single volume, No. 66-1.

Numerical machining data in the aforementioned six USAF Machinability Reports are scattered among descriptive material. In order to make these data more readily available, they were extracted and rearranged in tabular formats. The condition, microstructure, and hardness of each material are indicated. All the significant data for machining a given work material are presented in one horizontal line. This includes the tool material, tool geometry, cutting fluid, depth, width of cut, feed, tool life end point, and then the reamer life vs. cutting speed relationships. Thus, by scanning across a single line, one can obtain all pertinent machining data covering an actual test.

All the actual test data carried out on each work material are included. The test data were developed under closely simulated conditions which permit their direct application in the machining industry. The tool life end point referenced in the data sheets is a measure of the flank tool wear. Data are shown for high speed steel tools, cast alloy tools, and various grades of carbide tools. By scanning the lines of data for any specific work material, one can visually determine the relative tool life-cutting speed relationships for tool materials including tool geometry, feeds, and cutting fluids.

For any specific work material in a heat treated form and hardness, the recommended starting conditions are also given. The Recommended Starting Cutting Speed is indicated by the letter "R" for a given work material, and the line on which the "R" appears gives a resumé of the cutting parameters as well as the tool life. This recommended speed and its accompanying cutting conditions are not necessarily the optimum but rather serve to provide the user with suggested conditions which will give satisfactory tool life under normal shop conditions.

The optimum cutting parameters for machining a given work material on a given machine tool can be determined by obtaining tool life-speed, feed and cutting parameter relationships, and then using these data to calculate the cost and production rate for the various cutting parameter combinations. The optimum condition is that which produces minimum cost or maximum production. The machining test data which we present in this report are well suited for calculating optimum costs and production rates. This can be done by placing these actual machining test data into a computer which has been programmed to calculate costs and production rates. The computer output can be examined either graphically or in tabular form to ascertain the minimum cost or maximum production. N/C programmers need specific machining data in a form suitable for rapid and accurate inclusion in their machine tool programs. The data presented herein can either be scanned visually or computer optimized for this purpose. Any further questions regarding optimization should be directed as specific inquiries to the Air Force Machinability Data Center.

Supplementary data and information which will be useful for ready reference purposes are included in the Appendix, pages A-2 through A-18.

The six volumes from which these machining data were collected are:

	•	•	*
		•	
,		ь	

SUMMARY OF CONTENT

1	USAF	Machinability Report,	
	Vol.	1, (1950)	

Machining fundamentals and machining data on high temperature alloys.

2 USAF Machinability Report, Vol. 2, (1951) Turning data on 12 steels in various microstructural forms.

3 USAF Machinability Report, Vol. 3, (1954) Machining of titanium alloys including turning, milling, drilling, tapping, and grinding data.

4 USAF Machinability Report, Vol. 4, (1960) Machining of thermal resistant and high strength steels using a variety of important machining operations.

5 "Final Report on Machining of Refractory Materials", ASD-TDR-63-581, July 1963 Machining data on alloys of columbium, molybdenum, tungsten, tantalum and important high temperature alloys and nonmetallic materials; Tornetic drilling and tapping, high speed milling.

6 "Final Report on Machinability of Materials", AFML-TR-65-444, January 1966 Machining data on ultra-high strength steels, titanium alloys, and nickel and cobalt base high temperature alloys. A section is included on evaluation of Surface Integrity in machined and ground aerospace alloys.

NOTES:

- 1. Volumes 1 and 2, USAF Machinability Reports, are out of print but may possibly be obtained from libraries or on loan from AFMDC for a reasonable time or for reproduction.
- 2. A limited number of Volumes 3 and 4, USAF Machinability Reports, may be obtained from AFMDC without charge by providing information on specific needs for machining data on titanium, high strength steels, and high temperature alloys.
- 3. "Final Report on Machining of Refractory Materials", and "Final Report on Machinability of Materials" may also be obtained from AFMDC by providing need-to-know justification for the materials covered in each report (see description of content in Items 5 and 6 above).

PEANING STRENG THE DESCRIPTION

THE FUND OF THE STATE

10° 600

27、自然 新特代 等1**次**

REAMING

11.5

. 4...

The state of the s

· 二十二十二年 (第二十二年) 2014年 李明月(6月) 14年)

REAMING - INDEX TO DATA TABLES

ULTRA-HIGH STRENGTH S			-																											
DEAC	<i>A_p</i>	٠.				•	•	•	•		•	•	•	•	•	•				•		•	•	•	•	•			•	22
250 GRADE MARAGII	VG	ST	EEL		٠.	•		•			•	•	•	•		•				•		•		•		•		•		22
300 GRADE MARAGII	VG	ST	EEL		•	•			•	•	•		•		•	•						•			•		•			22
HP 9-4-25	•	•		•	•	•	•	•	•	•	•	•	•	•	•	•	•		•	•	•	٠	•	•	•	•	•	•	•	22
TIYANIUU ALLOYS																														
TI-BAL-1M0-1V .	•			•	•	•	•	•	•	•	•	•		•		•			٠	•			•	•			•		•	22
T1-3AL-13V-11CR	•	•		•	•		•	•	•	•	•	•	•	•	•	•	•		•	•	•		•	•	•	•	•	•	•	22
HIGH TEMPERATURE ALLO)42	-	HI	CK	EL	11	121	E 1	IR ()U(BHT	1																		
INCONEL 718		•		•	•		•	•	•		•	•		•				•	•	•	•	•	•		•			•	•	22
WASPALOY					•		•	•	•	•	•	•		•		•	•		•	•	•		•			•	•	•		229
RENE* .41	•	•		•	•	•	•		÷	•	•	•		•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	229
REFRACTORY ALLOYS - N	IOL	YB	DEN	UM	A	LLO)YS	7																						
MOLY-TZM		•							•		•	•	•			•		•						•			•	•		231
MOLY-0.5% TI	•	•		•	•	•	•	•	•	•	•	•	•		•	•	•	• '	•	•	•	•	•	•	•	•	•	•	•	231
REFRACTORY ALLOYS - C	400	-					.116																							
D-31	•	•			•		•	•		•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	٠	•	232
REFRACTORY ALLOYS - 1	AN	TA	LUM	A	LLI	DYS	3																							
90TA-10W																								L						231

				REAL	ER DI	REAMER DESCRIPTION	3	=	TOOL GEOMETRY		-IM	crack			Ē		REME	1 E	١	
MATERIAL	CONDITION	*	1001	TOOL MATL. DIA	=	NO. 0F	etvic	חנווא	CUANCED	0	FLUTB	ALLON.		FEED	LIFE		NO. OF HOLES	OF HOLE	S	
,	MICPOSTRUCTURE		TADE	TRADE INDUS- TRY NAME GRADE	in.	FLUTES	3116	RAND	COMMITTER		Appendix dix p. A-2	04 DIA. in.	HOLE in.	ipr	POINT in.	S T Y	SPEED-feet/minute R-Recommended Speed Appendix Page A-2	nded	Speed A-2	
ULTRA-HIGH STRENGTH DBAC	STEELS QUENCHED & TEMPERED TEMPERED MARTENSITE	56R _C	883	C-2	.272	4	CHUCKING	°0 #	45° x .080"	01	53	.020	.5 THRU	.001	.012	5	80	25	15	4 9
E	E	E	E	E	E	=	CHUCKING -5 x .010 THONED ON CORNERS	E E	E	E	E	=	E	. 002	E	20	47	09 88	35	35
E	E	=	=	E	=	E	CHUCKING	E E	E	E	Ė	E	E	E	E			75 S.		
DGAC	QUENCHED & TEMPERED TEMPERED MARTENSITE	56Rc	883	C-2	272.	4	CHUCKING -5 x .010* HONED ON CORNERS	O WH	45° x . 060"	10	53	.020	.5 THRU	.002	.012			65		
DGAC	QUENCHED & TEMPERED TEMPERED MARTENSITE	56R _C	883	C-2	.272	+	CHUCKING	0°0	45° x . 060"	10	53	. 020	.5 THRU	.002	.012			65		
(E	ı E	E	E	E	E	E	ε	E E	E	E	1:20	E	ı	E	E			30		
250 GRADE Maraging Steel	ANNEALED MARTENSITE	321	•	M2 HSS	272.	9	CHUCKING	O RH	45° x .060"	7	52	. 022	.5 THRU	. 009	.006	40	123	170 160 R		
r	E E	E	•	E	E	E	E	E E	E.	E	53		ш	E	=	13	175	190		
E	E E	E	•	E	E	E	E	E E	E	Е	1:20	E	E	E	E	35	100	85		

				REALE	R 06.	REAMER DESCRIPTION	3		TOOL GEOMETRY	 <u>-</u>	ET.	S TOPE			131		3	R LIFE	
MATERIAL	CONDITION	#	1901	TOOL MATL. DIE NO. OF	2	0. OF	eTVI E	HEI 17	CHAMEED	0 130			LENGTH	FEED	LIFE		8	NO. OF HOLES	S
	MICROSTRUCTURE		TRADE	TRADE TRY NAME GRADE	, ii	LUTES		HAND		1	Appendix dix 0. A-2	8	를 .:	اق.	20.87		SPEED-fe R-Recomme Appendix	ER	Speed A-2
ULTRA-HIGH STRENGTH STEELS - (cont.) 250 GRADE ANNEALED & MARAGE MARAGING STEEL MARTENSITE	0	50R _C		HSS H	272.	ω	CHUCKING	O RH	45 x . 060"	_	52	. 022	.5 THRU	. 005	900.	00 00	35 1	170	
Ξ	= =	=	1	M33 HSS	ï	=	E	E E	E	=	=	E	E	E	Ľ	130	001	06 001	
E	E E	=	•	H SS	=	=	E	ı ı	E	(E	53	u	E	и	ŧ	30	45	271	
250 GRADE Maraging Steel	ANNEALED & MARAGED MARAGED MARTENSITE	50R _C		M2 HSS	272	ဖ	CHUCKING	О	45° x .060"	7	52	.022	. 5 THRU	.005	900 .	170			
E		l t		ŧ	Ē	=	in :	E E	i i	l E		E	E	600.		120			
=	E E	=	•	*	=	=	a	E E	E	E	E.	¥	E	.015	E	70			
300 GRADE Maraging Steel	ANNEALED MARTENSITE	348	•	H SS	272.	9	CHUCKING	ВН	45° x . 060"	7	52	.022	.5 THRU	600°	900	65 1	120 3	300 75	
E	E		•	*	=	=	=		E	=	53	н	E	ī.	и	80 1	100	160	
E	Ε Ε	E	ı	E	=	=	:	£ £	E	E	1:20	E	£	E	E	35	09		

		. B.										
3	LES	inuted Spec	8 2								l	
REAMER LIFE	足り	feet/r mende ix Pa	45	120 35	140	140	85			145	120	
REA	₽.	SPEED-feet/minute R=Recommended Speed Appendix Page A-7	37	60	30			30	30	58	110	
		S = 4	35	40 55	16					15	125	
100L	LIFE		900.	=	E	900.	E	900.	E	900.	=	
		ë	.005	E	£	. 005	600.	.005	600.	600.	E	
1200	LENGIH	in.	5 THRU	r	E	.5 THRU	/E	5 THRU	E	5 THRU	E	
STOCK		₹ <u>-</u> .:	. 022	E	E	. 022	E	. 022	£	. 022	E	
—				_							0	
CUT-		₹ 6	. 25	53	52	52	E	52	<u> </u>	52	1:20	
_	0 130	1	1	=	=	7	į.	1	E	7	E	
TOOL GEOMETRY	CHANEED		45°x .060"	E	=	45° x .060"	E	45° x.060°	E	45°x .060"	£	
흔	HELLX	HAND	°0 #2	E E	E E	O RH	E =	00 HR	(E E	0 0 HR	E E	
No	CTVIE	=	CHUCKING	E	E	CHUCKING	E	CHUCKING	=	CHUCKING	E	
REAMER DESCRIPTION	NO. 0F	LUTES	6	=	E	æ	E	9	=	9	E	
ER DE	=	į	272.	=	E	272	E	272.	E	.272	E	
REAN	ATL.	INDUS- TRY GRADF	M33 HSS	H SS	=	M2 HSS	=	#2 HSS	E	M2 HSS	E	
	TOOL MATL.	TRADE INDUS- TRY NAME GRADE		1	•	•		•	•	•	•	
	HE		52Rc	ŧ	E	52Rc	- #	52R _C	E	341	£	
	CONDITION	MICROSTRUCTURE	STEELS - (cont.) ANNEALED & MARAGED MARTENSITE	= =	E E	ANNEALED & MARAGED MARAGED MARTENSITE	E E	ANNEALED & MARAGED MARTENSITE	E E	ANNEALED SPHEROIDIZED	E E	
	MATERIAL		ULTRA-HIGH STRENGTH STEELS 300 GRADE MARAGING STEEL MARTEN	E	E	300 GRADE Maraging Steel	ш	300 GRADE Maraging Steel	ı	НР 9-4-25	£	

					1				1000 0000		Ting	E E	I CHETT	ננט	3				ع د
0)		No.	TOOL MATL.	ITL.	=	0. OF	OTV! C	ATION	COUNTRO		FIE	ALLOW.		3	LIFE		.		2
1	MICROSTRUCTURE		RADE	1 1 4	₫ .5	FLUTES	SI LE	HAND		<u>.</u>	Appendix	8 2 .5	. <u>F</u>	. =		SI-	SPEED-feet/minute R-Recommended Spee		et/minute med Speed
		†	1		+	\dagger													1
	ANNEALED	302	,	1112	272.	ဖ	CHUCKING	°.	45° X .060"	7	53	.022	.S THRU	. 009	900.	55 55	120	300	_
						,		#								3	3	₹	
	=			1				E				1				20	105	200	
	E	=	•	•		•	=	E	E	E	£	•		E	:	2	70	85	
E	E	E			=	E	2	E	E		11					2	65	300	
	E							E			1:20			:		5	65	00	
	SOLUTION TREATED & AGED			5		31		%								*0	37	25	75
T1-3AL-13V-11CR BETA		2		S	.272	ø	CHUCKING	£	45 × .060	9	ი ი	.020	.5 THRU	ē	. 015	70	20	04	30
	u				-			E	,	1					:	2	35	95	
=	E	=	• .	E	E	E	Ē	t t	E .	=	E	E	E	Z00°	E	0,	20	30	
, E	н						1	н	2			,			1	ဖ	06	185	-
:	E		,	<u>.</u>		1	=	E	:	=	•	=	=	. 005	F	2	S	8 €	
HIGH TEMPERATURE ALLOYS	S - NICKEL							%								, u	•	-	_
	AUSTENITIC	245	1	HSS.	.272	9	CHUCKING	£	45° x .060"	7	25	. 022	.5 THRU	.005	900.	200	20	+	-
	E					•		E		11	•		,			35	75	1	
	ц		993	7-7	:	•	:	E	= 1	=	=	=	E	600	E	06	70 R		
										·									

				REAMER	PESC	REAMER DESCRIPTION	4	10	TOOL GEOMETRY	_	CUT-	STOCK	i i i		1001	REAMER LIFE	IFE
MATERIAL	3	BHN.	LOOL NA	TL.	2	NO. 0F	CTVIE	7	CUANCED	0 130	FLUE	ALLOW.	LENGTH		III.	NO. OF ED	LES
	MICROSTRUCTURE		TRADE IN	TRADE INDUS-UN. IN. NAME GRADE IN.	<u>.</u> .	JTES	1 1 1	RAND HAND	COAMICE	1	Appendix dix D. A-2	₹ <u>.</u>	HOLE in.	īdi	POINT in.	SPEED-feet/minute R=Recommended Speed Appendix Page A-2	minute d Speed ge A-2
HIGH TEMPERATURE ALLOYS - NICKEL BASE WROUGHT - SOLUTION TREF (cont.) WASPALOY	NTEO	293	* * * * * * * * * * * * * * * * * * *	M2 HSS -2	. 272	9	CHUCKING	O BH	45° x .060"	7	52	.022	.5 THRU	.002	900.	45	
	Е Е	E	•			±	ш	= =	E	ı.	41	ш	#	.005	E	80	
±	E E	=	•	E		=	#	= =	E.	ži.	ш	±	E	600.	=	80 45 80	
WASPALOY	SOLUTION TREATED AUSTENITIC	293	-	M2 HSS -2	272.	9	CHUCKING	°0 RH	45° x .060"	7	52	. 022	.5 THRU	.002	900.	15	
±	= =	E	•	E .		=	ŧ		ı.	E	14	ŧ	E	. 005	E	60	
	E E	E	•	E		E	ŧ	: :	E	E	ı.	E	E	. 009	Ε	10	
RENE" 41	SOLUTION TREATED AUSTENITIC	321	,	M2 HSS .272		9	CHUCKING	. O HR	45° x .060"	10	53	. 020	. 5 THRU	. 005	910.	22 95 72 30 25 20	65
п	E E	=	•	±	E .	E	E	E E	E.	z.	52	E	Ė	t t	ŧ	30	
	E E	=	•	=	=	=	E	: :	ŧ	E	11 1: 10	E	B		E l	8 8	
			911														

				REAL	30 83	REAMER DESCRIPTION			TOOL SECRETRY		3	STEELS.	8		TOD		REMER LIFE	ببر
MATERIAL	3	2	100L	MIL		10. OF	ervi e	A1 137	ST THE CO.	0					LIFE	=	B. F. BEL	23
	MICROSTRUCTURE		TRADE TRY NAME GRADE in.	INDUS- TRY GRADE	<u> </u>	FLUTES	3116	HAND			Appendix	5 5 6	ë ge	-		2945 -03345		Speed Speed
BASE WROUGHT - SOLUTION TRE (cont.) RENE: 41	SOLUTION TREATED AUSTENITIC	321	,	HZ HSS	272.	ω	CHUCKING	°0 ##	45° x .060°	2	53	.020	.5 THRU	. 005	910.		72 20	
E	EE	E	•	E	E	2	E	E E	r	E	F	E	Е	600.	=		32	
E	E E	E	1	E	E	E	E	E E	E	E	.	E	E	.015	E		7 20	
RENE* 41	SOLUTION TREATED & AGED AUSTENITIC	365	•	IK2 HSS	.272	æ	CHUCKING	° O HH	45° x .060"	10	53	.020	.5 THRU	. 005	910.	30 2	40 94 25 20	71
E	E E	E	•	E	E	E	E	E E	E	E	52	E	E	E	E)			17
E		=	•	и	£	E	E	E E	E	E	11	£	E	E	E			17
RENE" 41	SOLUTION TREATED & AGED AUSTENITIC	365	•	M2 HSS .	272	9	CHUCKING	O RH	45° x .060"	10	53	.020	.5 THRU	. 005	.016		94	
E	Е, Е,	±	•	E .	E	t	E	E E	Ľ	E	E	E	E	600.	Ė		35	
E	E	E	•	F	E	E	E	E E	E	E	E	E	E	.015	E		15	
		_																

				REA	ER DE	REAMER DESCRIPTION	8	=	TOOL GEOMETRY		-IR3	PTOFF	_		TOUL		REAN	FR 1.16	4
MATEDIAL	CONDITION	3	T00L	TOOL MATL. B.		NO. OF		21	2	i	TING FILE	ALLOW.	LENGTH	FEED	LE		₽.	NO. OF HOLES	E
	MICROSTRUCTURE		TRADE	TRADE INDUS- NAME GRADE	<u>.</u>	FLUTES		RELIA RAND	CHAMPER	ב ב	Appendix			. <u>.</u>	PO INT	E E	Recomm	SPEED-feet/minute R=Recommended Speed	Speed Speed
REFRACTORY ALLOYS	EXTRUDED ELONGATED GRAINS	229	•	M2 HSS	272.	g	CHUCKING	° H	45° x .060"	<u> </u>	53	. 020	.5 THRU	600.	.012	105	36	45 62 R	
E	E	=	•	=	Ė	a- pa-	E	E E	E	=	£	5	E	.015	E	105	85	52	
E	EE	E	•	E	=	E	=	E E	=	E	=	H	E	.020	E			25	
MOLY-TZM	EXTRUDED ELONGATED GRAINS	229	•	M2 HSS	272.	9	CHUCKING	O °	45°× .060"	10	1:20	.020	.5 THRU	.015	.012		15		
п	E E	2	•	E	=	=	=		#	E	52	14	E	E	ŧ.		30		
н		11	•	E	=	н	u	ш	Ħ	=	53	u	t.	н	E		40		
MOLY-0.5% TI	STRESS RELIEVED ELONGATED GRAINS	220	4	M2 HSS	.213	Ф	CHUCKING	10°	45° x .060"	10	53	020	.5 THRU	.000	.010		25		
н	E E	ŧ	11	ŧ	E	ı.	E	E E	H	=	E	н	E	.011	ε		35		
£	E =	=	ı	E	Ŀ	±	Ε	E E	**	E .	E	E	4 -	.015	E		45 R		
E	E E	±		=	=	E	=	E E	E	=	£	(E	ŧ	. 920	E		85		

MATERIAL				REAL	ER DE	REALER DESCRIPTION	5	12	TOOL GEOMETRY	¥					151			LIFE	
	CONDITION	3	TOOL MATL.	111.		10, OF	3 174	200				=		2	_			2	
	MICROSTRUCTURE		TRADE INDUS-	TRY		FLUTES	STALE	HAND HAND						ë			SPEED-feet/sinute	Cainte Se Se S	# 1 2
fcont.) MOLY-0.5% T:	- MOLYBDENUM ALLOYS STRESS RELIEVED ELONGATED GRAINS	220	,	HSS .	.213	ø	CHUCKING	° =	45° к .060	2	53	.020	S THRU	600	010.		-		
E	E =	E		E	=	E	E	00 #	Ē		2	E	(8)	E	E	23			
E	= =	ŧ	,	E	E	2	E	°0 = =	Ε	¥	=	F	£	=	£	30			
MOLY-0.55 TI	STRESS RELIEVED ELONGATED GRAINS	220		M2 HSS -	.213	9	CHUCKING	10°	45° x .060"	01	1:20	. 020	.5 THRU	.020	010.	25.			
ŧ	E E	E	•	E	i E	н	E	E E	E		52	£	E	E	E	59			
и	E E	TE.	6	ŧ	=	ŧ	E	E E	E	E	53	E	ε		E	60			
REFRACTORY ALLOYS D-31	- COLUMBIUM ALLOYS EXTRUDED & STRESS RELIEVED ELONGATED GRAINS	207	1	HSS .	.213	9	CHUCKING	10°	45° x .060"	0	52	.020	.5 THRU	. 005	.012	25 40	40 54 62 150	103 125 R	
ш	E E	£		lle.	E	u	E	E E	E	E .	53	E .	E	E	E		12		
н	E	E	•	E	E	E	ш	E E	E	=	11	E	E	E	E			14	
E	E E	Ε	•	E	E	E	E	0 NH RH	E	ŧ	52	E	E	E	E		150	,	

				REAM	ER DE	REAMER DESCRIPTION	NC.		TOOL GEOMETRY	≩	-tno			_	Ī	_	REAMER	LIFE	
MATERIAL	CONDITION	BEN SE	T001	MTL.	=	NO. OF	CTVIE	HE IX	CHANEED	DEI 0		₹	LENGTH		LIFE		NO. OF HOLES	HOLES	
	MICROSTRUCTURE		RADE KANE	TRADE TRY NAME GRADE in.	<u>.</u> .e.	LUTES	1	HAND	COMPLE	Į	Appendix dix p. A-2	8 <u>2</u> :	in.	ip	2 E :		EED-fee ecommen	SPEED-feet/minute R=Recommended Speed Appendix Page A-2	a B ~
REFRACTORY ALLOYS - (cont.)	EXTRUDED & STRESS RELIEVED ELONGATED GRAINS	207	•	H SS	.213	9	CHUCKING	10°	45° x .060"	01	52	.020	.5 THRU	.002	.012		33	33	
E	E E	/E	,	=	E	E	=	E E	E	=	E	F	=	.005	E		54	54	
E	E E	=		=	=	E.	£	E E	E	=	=	:	E	600°	E		12	12 50	
REFRACTORY ALLOYS -	AS FORGED EQUIAXED GRAINS	207	1	HSS .	.213	9	CHUCKING	10°	45° x .060"	10	53	.020	.5 THRU	. 005	.012		30	30	
E	E E	=	•	E	ŧ	E	ш	E E	E	E	u	E	ı.	600.	E	100	67 64 85 75	64 42 75 50	
#	E E	Ė		E .	E	u	E	£ £	E	=	i.	E	E	.015	E		,	5 27	
90TA-10W	AS FORGED EQUIAXED GRAINS	207	,	M2 HSS -:	213	9	CHUCKING	10°	45° x .060"	01	52	.020	.5 THRU	600.	.012			75	
=	E E	H	•	E	E	#	ш	E E	5	E	1:20	E	£	E	E		1	4 75	
E	E E	E		E	=	E	E	E E	=	E	53	E	E	E	E		7	62	
3									μ.										

APPENDIX

CUTTING FLUIDS USED IN TESTS

CODE NO.	DESCRIPTION
00	None, Dry
11	Water soluble oil - light duty
13	Heavy duty soluble oil
21	Chemical emulsion
31	Sulfurized mineral lard oil - light duty
52	Sulfo-chlorinated mineral lard oil - medium duty
53	Highly chlorinated mineral oil - heavy duty
54	Heavy duty highly chlorinated cutting oil
83	Ti-Kut Oil (especially for titanium)
mc	Mist through cutter

Solution Concentrations:

1:10 denotes one part of concentrate to 10 parts of water by volume

1:15 denotes one part of concentrate to 15 parts of water by volume

1:20 denotes one part of concentrate to 20 parts of water by volume

1:25 denotes one part of concentrate to 25 parts of water by volume

R = RECOMMENDED CUTTING SPEED

The Recommended Starting Cutting Speed is indicated by the letter "R" for a given work material, and the line on which the "R" appears gives a resume of the cutting parameters as well as the tool life. This recommended speed and its accompanying cutting conditions are <u>not</u> necessarily the optimum but rather serve to provide the user with suggested conditions which will give satisfactory tool life under normal shop conditions. To take full advantage of the data presented, it will be necessary to optimize; i.e. to determine maximum production and minimum cost. (See general comments on optimizing in the Introduction and/or direct specific inquiries concerning optimization to the Air Force Machinability Data Center.)

BLANK PAGE

CARBIDE GRADE CHART

C-1 to C-8 MACHINING APPLICATIONS

							MACHINING AF	PLICATION
CARBIDE				INDUSTRY	CODE			
MANUFACTURERS	C-1	C-2	C-3	C-4	C-5	C-8	C-7	C-8
ADAMAS	В	A AM PWX	PWX	AAA	DD 5X 434	6 X	7X C 548 Titan 80*	CC Titan 80
AMCARD		D15 D13			••			••
BESLY-WELLES	B101	8106 8188	B108	8211	8109 8221	8 102	8103 8104 8205 8245	8207 8365*
CARBOLOY	448	883 860	883 905 895	999 995 320	370 768	370 788 78 350	350 78 320	320
CARMET	CA-3	CA-4 CA-443	CA-7	CA-8	CA-610 CA-740	CA-608 CA-720	CA-711	CA-704
COROMANT	H20	H13 H1P	HIP	H05	\$8 \$4	\$2	SIP	F02*
FIRTH-LOACH	FA-5	FA-8	FA-7	FA-8	FT-3 FT-4 FT-5	FT-5 FT-62	FT-6 FT-62	FT-7 FT-72*
FIRTH STERLING	Н	HA H-23	HE	HF	TO4 NTA	TXH T22	T22 TXL	T31 WF*
FUTURMILL		DMC21	••		DMC30 DMC32	DMC32	DMC35	••
KENNAMETAL	K1	K6 C8735 K68	K68 K8	K11	KM K21 K2S	K2S K3H K4H K45	K45 K5H K7H	K7H K165*
MULTI METALS	OM 1	OM2	OM3	OM4	4M5		••	
NEWCOMER	NIO	N20	N30	N40	N50	N60	N70 NM-93*	N60 NM-83* NM-85*
SINTERCAST	Ferro- Tic J	Ferro- Tic J			Forro- Tic J	Forro- Tic J		
SPEEDICUT MITIA	A	В	C	C	TA10 TA5	TTA	TE	TE
TALIDE	C-89	C-91	C-93	C-95	\$-880	S-901	\$-92 \$-900	S-94
TUNGSTEN ALLOY	9	9H	90	98	117 9\$ 10T	9\$ 10T 5\$	8T 5S	5\$
UNIMET	V10	U20	U30	U40	U53	U53 U60	U70 U73	U73 U80 U88*
VALENITE	VC-1	VC-2 VC-22 VC-28	VC-3	VC-4	VC-125 VC-55	VC-125 VC-8	VC-7	VC-8 VC-83* VC-85*
VR/WESSON	2A-88 VR-54	2A-5 VR-54	2 A -7	VR-52 2A-7 VR-85*	WS VR-77 VR-89 VR-75	VR-75	VR-73 WH HV VR-85*	HV VR-73 VR-65*
WALMET	WA-141 WA-1 WA-159	WA-2 WA-63 WA-149	WA-35 WA-3	WA-4	WA-68 WA-5	WA-5 WA-6	WA-147 WA-7	WA-8
WENDT-SONIS	CQ12	CQ2	C Q 3	CQ4	CY 1 2 CY 16	CY 16 CY 5	CY14 CY2 T18*	CY31 T18*
WICKALOY	N	н	нн	нин	X7 A X7	88	GX	FX
WILLEY'S	E8	E6	E5	E3	945 8Å 10Å	8.4	606 6 A	6 A X 509

CAST IRON, NON-FERROUS AND NON-METALLIC MATERIALS

C-1 C-2 C-3 C-4

Roughing General Purpose Finishing Precision Finishing

STEEL AND STEEL ALLOYS

C-5 C-6 C-7 C-8

Roughing General Purpose Finishing Precision Finishing

Listings do not necessarily imply equivalency of various manufacturer's grades.
This chart is not to be considered an endorsement of or an approved list of any manufacturer's products. Grades containing more than 50% Titanium Carbide.

IDENTIFICATION AND TYPE CLASSIFICATION OF HIGH SPEED TOOL STEELS

					SYMBO	M, MOLYBD	ENUM TYPES
TYPE	I DENT	IFYIN	G ELEM	ENTS,	IN PE	RCENT	APPLICATION
	C	W	Mo	Cr	٧	Co	
M1	.80	1.50	8.00	4.00	1.00	•	GENERAL PURPOSE
M2	.85	6.00	5.00	4.00	2.00	-	GENERAL PURPOSE
M3 CLASS 1	1.05	6.00	5.00	4.00	2.40	•	FINE EDGE TOOLS
43 CLASS 2	1.20	6.00	5.00	4.00	3.00	•	FINE EDGE TOOLS
M4	1.30	5.50	4.50	4.00	4.00	•	ABRASION RESISTANT
M6	. 60	4.00	5.00	4.00	1.50	12.00	HEAVY CUTS - ABRASION RESISTANT
M7	1.00	1.75	8.75	4.00	2.00	-	FINE EDGE TOOLS - ABRASION RESISTANT
M10	. 90		8.00	4.00	2.00	- Cr	GENERAL PURPOSE - HIGH STRENGTH
M15	1.50	8.50	3,50	4.00	5.00	5.00	HEAVY CUTS - ABRASION RESISTANT
M30	. 80	2.00	8.00	4.00	1.25	5.00	HEAVY CUTS - ABRASION RESISTANT
M33	. 90	1.50	9.50	4.00	1,15	8.00	HEAVY CUTS - ABRASION RESISTANT
M34	. 90	2.00	8.00	4.00	2.00	8.00	HEAVY CUTS - ABRASION RESISTANT
M35	. 80	6.00	5.00	4.00	2.00	5.00	HEAVY CUTS - ABRASION RESISTANT
M38	. 80	6.00	5.00	4.00	2.00	8.00	HEAVY CUTS - ABRASION RESISTANT
M41	141 1.10 6.75 3.75 4.25 2.00 5.00		5.00	HEAVY CUTS - ABRASION RESISTANT			
M42	1.10	1.50	9.50	3.75	1,15	8.00	HEAVY CUTS - ABRASION RESISTANT
M43	1.25	1.75	6.75	3.75	2.00	8.25	HEAVY CUTS - ABRASION RESISTANT
M44	1.15	5. 25	6.25	4.25	2.25	12.00	HEAVY CUTS - ABRASION RESISTANT
					SYMBO	L T, TUNGS1	TEN TYPES
T1	.70	18.00		4.00	1.00	•	GENERAL PURPOSE
T2	. 80	18.00		4.00	2.00		GENERAL PURPOSE - HIGHER STRENGTH
T4	.75	18.00	•	4.00	1.00	5.00	HEAVY CUTS
T5		18.00	•	4.00	2.00	8.00	HEAVY CUTS - ABRASION RESISTANT
T6		20.00	•	4.50	1.50	12.00	HEAVY CUTS - HARD MATERIAL
T7	. 75	14.00	-	4.00	2.00	•	PLANER TOOLS
T8	. 75	14.00		4.00	2.00	5.00	GENERAL PURPOSE - HARD MATERIAL
T9		18.00	-	4.00	4.00	•	EXTREME ABRASION RESISTANT
T15	1.50	12.00	-	4.00	5.00	5.00	EXTREME ABRASION RESISTANT

GENERALLY ALL OF THE ABOVE HIGH SPEED STEELS ARE MANUFACTURED BY THE FOLLOWING COMPANIES:

ALLEGHENY LUDLUM STEEL CORPORATION

BETHLEHEM STEEL CORPORATION

BRAEBURN ALLOY STEEL DIVISION, CONTINENTAL COPPER & STEEL INDUSTRIES, INC.

THE CARPENTER STEEL COMPANY

COLUMBIA TOOL STEEL COMPANY

CRUCIBLE STEEL COMPANY OF AMERICA

FIRTH STERLING, INC.

JESSOP STEEL COMPANY

LATROBE STEEL COMPANY

H. K. PORTER COMPANY, INC., VULCAN-KIDD STEEL DIVISION

SIMONDS SAW AND STEEL COMPANY

UNIVERSAL-CYCLOPS STEEL CORPORATION

VANADIUM-ALLOYS STEEL COMPANY. DIVISION OF VASCO METALS CORPORATION

HIGH SPEED STEELS M41 THROUGH M44 ARE MADE BY:

M41 - CRUCIBLE STEEL COMPANY OF AMERICA

M42 - VANADIUM-ALLOYS STEEL COMPANY, DIVISION OF VASCO METALS CORPORATION

M43 - LATROBF STEEL COMPANY

M44 - BRAEBURN ALLOY STEEL DIVISION, CONTINENTAL COPPER & STEEL INDUSTRIES, INC.

This chart is not to be considered an endorsement of any manufacturer's product or an approved list of any manufacturer's products.

The state of the s

CAST ALLOY TOOL MATERIALS

Nominal Composition

Trade Name	<u>% Co</u>	% Cr	% W	Hardness R _C
Blackalloy 525	44.0	24.0	20.0	63 - 64
Blackalloy T. X. 90	42.0	24.0	22.0	66 - 67
Crobalt #1	48.0	30.0	14.0	59 - 60
Crobalt #2	40.0	33.0	18.0	61 - 62
Crobalt #3	40.0	33.0	20.0	63 - 64
Stellite 98M2	38.0	30.0	18.5	63
Stellite Star J	43.0	32.5	17.5	61
Stellite 3	50.0	31.0	12.5	59
Stellite 19	53.0	31.0	10.5	55
Tantung G	47.0	30.0	15.0	60 - 63
Tantung 144	45.0	28.0	18.0	62 - 64.5

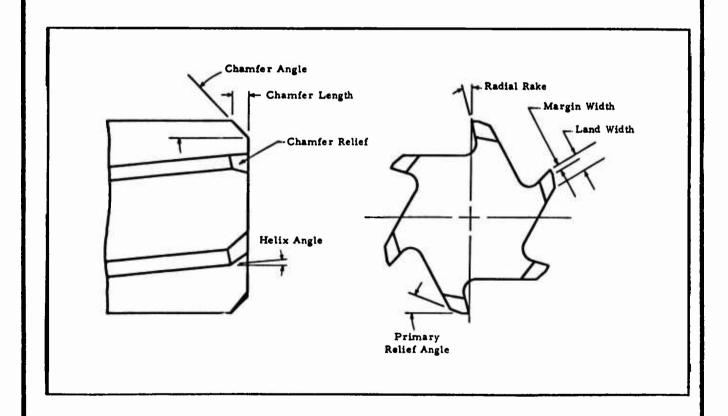
Trade Name Manufacturer

Blackalloy	Blackalloy Company of America
Crobalt	Crobalt, Inc.
Stellite	Union Carbide Corp., Stellite Division
Tantung	VR/Wesson Company

The procedure used for making cast alloy tools is to melt the particular analysis in an electric furnace and then cast the tools to shape with small stock allowance for finish grinding. The tools are at maximum hardness when removed from the molds and cannot be hot or cold worked and do not respond to heat treatment.

This chart is not to be considered an endorsement of any manufacturer's product or an approved list of any manufacturer's products.

TOOL NOMENCLATURE - Reaming



ROCKWELL-BRINELL - ULTIMATE TENSILE STRENGTH HARDNESS CONVERSION CHART

ROCKWELL C to	BRINELL 3000 KG.	ROCKWELL B	to BRINELL 500	and 3000 KG.
FOR HARDENED	STEEL AND ALLOYS		TEEL, STEEL OF SOF RON AND MOST NONFE	T TEMPER, GRAY AND
ROCKWELL C 150 Kg. Load "Braie"	BRINELL 3000 Kg. Load 10 mm Ball	ROCKWELL B 100 Kg. Load 1/16" Dia. Bali	BRINELL 500 Kg. Load 10 mm Ball	BRINELL 3000 Kg. Load 10 mm Ball
80 59 58 57 56	614 600 587 573 560	100 99 98 97 96	201 195 189 184 179	240 234 228 222 216
55 54 53 52 51	547 534 522 509 496	95 94 93 92 91	175 171 167 163 160	210 205 200 195 190
50 48 48 47 46	484 472 480 448 437	90 89 88 87 86	157 154 151 148 145	185 180 176 172 169
45 44 42 40 38	426 415 393 372 352	85 84 83 82 81	142 140 137 135 133	165 162 159 156 153
36 34 32 30 28	332 313 297 283 270	80 79 78 77 76	130 128 126 124 122	150 147 144 141 139
26 24 22 20	260 250 240 230	75 74 72 70 68	120 118 114 110 107	137 135 130 125 121
	3000 KG. Ensile Strength Teels	66 64 62 60 58	104 101 98 95 92	117 114 110 107 104
BRINELL 3000 Kg. Load 10 mm Ball	ULTIMATE TENSILE Strength, psi	56 54 52 50 48	90 87 85 83 81	101 - - - -
200 225 250 275	100,000 108,000 122,000 141,000	46 44 42 40 38	79 78 76 74 73	
300 325 350 375	158,000 174,000 188,000 202,000	36 34 32 30 28	71 70 68 67 66	
400 425 450 475	215,000 227,000 238,000 249,000	24 20 16 12 8	64 62 60 58 56	
500 525 550 575 670	258,000 267,000 282,000 295,000 308,000	0	55 53	

DEFINITIONS

METALLURGICAL AND HEAT TREATING TERMS

ACICULAR Alloyed cast irons, which develop high strength

on cooling and which have a structure containing martensite mixed with other microconstituents,

are termed "acicular," i.e. needlelike.

AGING A heat treatment of a previously solution heat

treated alloy at temperatures which permit solid state precipitation to occur. This treatment usually develops an increase in the hardness and strength of an alloy but may decrease machina-

bility.

ALLOY STEELS Steels with carbon from 0.1% to 1.1% and con-

taining alloying elements such as nickel, chromium, molybdenum, vanadium, etc. The total of all such alloying elements in these type steels

is usually less than 5%.

ALPHA TITANIUM

ALLOYS

See Titanium Alloys

ALPHA-BETA TITANIUM ALLOYS See Titanium Alloys

ANNEALING A heat treatment of an alloy above its critical

or recrystallization temperature, usually followed by furnace cooling. This treatment results in low hardness, high ductility, and usually improves tool life. Some alloys have too high a ductility after annealing and this may lead to poor finish and poor chip control in machining. For such alloys, the cold drawn condition is

preferred.

AUSTENITE A solution of one or more elements in face-

centered cubic iron. In most alloy steels, austenite is stable at heat treating temperatures and changes to other constituents such as pearlite, ferrite, martensite, etc. In the austenitic stainless steels, it is stable at room temperature

as a result of alloying.

BETA TITANIUM

ALLOYS

See Titanium Alloys

BRINELL HARDNESS A test for determining the hardness of a material by forcing a hard steel or carbide ball of specified diameter into it under a specified load. The result is expressed as the Brinell hardness number, which is the value obtained by dividing the applied load in kilograms by the surface area of the resulting impression in square millimeters. Standard tables are available which indicate the hardness for various size penetrators and indenter loads. The minimum thickness of the test piece should be 1/2". Indenters are spherical, and the load is usually 3000 kg or 500 kg for softer materials.

CARBIDES

Usually refers to the general class of pressed and sintered tungsten carbide cutting tools which contain tungsten carbide plus smaller amounts of titanium and tantalum carbides along with cobalt which acts as a binder. It is also used to describe hard compounds in steels and cast irons.

CARBURIZING

Carburizing heat treatments are used to increase the carbon content at the surface of steels. This is done by placing the parts in contact with carbonaceous solids, liquids, or gases at approximately 1600 to 1700°F. The depth of penetration and amount of carbon pickup are a function of the time, temperature and carburizing medium. Hardening treatments of carburized surfaces develop hardnesses up to 65 R₂.

CASE HARDENING

A heat treating method by which the surface layer of alloys is made substantially harder than the interior. Carburizing and nitriding are common ways of case hardening steels.

CAST STEELS

Steels which are cast to shape and used without having been hot rolled or forged but which are usually heat treated to produce annealed, normalized, or hardened products.

CERAMICS

A term used to describe the general class of hard, brittle and high melting nonmetallic materials such as aluminum oxide, zirconium oxide, beryllium oxide, etc. The same term is also used to designate aluminum oxide and other similar type ceramic cutting tools.

COLD DRAWING

A method of reducing hot rolled rod to size by pulling the rod through a die of the desired diameter. This is frequently done at room temperature but, in any event, must be carried out below the recrystallization temperature of the material. This process often improves machinability by providing better surface finish and chip control.

COLD ROLLING

A method of reducing hot rolled flat stock by passing it between rolls to the desired thickness. Temperature limitations are the same as for cold drawing.

CORROSION RESISTANT STEELS This is a term often used to describe the cast stainless steels and to differentiate those cast steels used primarily for corrosion applications.

DRAWING

A term often used to mean the same as tempering, which is the preferred designation. 'Drawing' should be reserved for hot and cold mechanical working operations.

DUCTILE IRONS

These irons are made by inoculating liquid iron from a cupola with magnesium-nickel or other inoculants to produce microstructures similar to those found in gray iron except that the graphite is in a nodular instead of a flake form. These cast irons are sometimes called "nodular irons."

FERRITE

A solid solution of one or more elements in bodycentered cubic iron. It is a common microstructure found in steels and cast irons and generally provides good tool life.

FORGED

A cold or hot mechanical working process performed by presses or hammers and used to shape alloys. Treatments such as annealing or normalizing should generally be applied after forging to improve machinability and uniformity thereof.

GRAY CAST IRONS

Alloys primarily of iron, carbon, and silicon along with other alloying elements in which the graphite is in flake form. These irons are characterized by low ductility but have many other properties such as good castability and good damping capacity.

Н

A letter listed along with the composition of aluminum and magnesium alloys indicating that the alloy has been cold worked. H12, H14, etc., are used to indicate various degrees of working.

HARDENING

Designates various heat treatments such as quench hardening, age hardening, and precipitation hardening.

HARDNESS

The ability of a metal to resist penetration. The maximum hardness obtainable in a steel is a function of the carbon contentand the heat treatment and is affected only slightly by the alloy content.

HEAT RESISTANT STEELS Cast steels which are highly alloyed and used for high temperature applications such as furnace conveyor parts, etc.

HIGH SPEED STEELS

Tool steels which contain tungsten, molybdenum, vanadium, cobalt, carbon, and other elements and which are widely used as tools for various machining processes.

HIGH TEMPERA-TURE ALLOYS Generally alloys of nickel, cobalt, or iron which are used at temperatures in excess of 1200°F in applications such as rockets, jet engine blades, and compressor and turbine discs, etc.

HOT ROLLED

A term used to describe alloys which are processed through hot rolls. Many alloys are often used in this state and sometimes machinability varies because of differences in cooling conditions from lot to lot.

LEADED ALLOYS

Alloys to which lead has been added to improve machinability.

MALLEABLE

Irons made by malleablizing white cast iron. See Malleablizing.

MALLEABLIZING

Process of annealing brittle white cast iron in such a way that the combined carbon is wholly or partly transformed to graphitic or temper carbon nodules in a ferritic or pearlitic microstructure, thus providing a ductile and machinable material.

MARAGING STEELS High strength steels of complex composition in which strength is achieved through aging of martensite.

MARTENSITE

An acicular or needlelike microstructure that is formed in quenched steels. It is very hard and brittle in the as quenched form and therefore is usually tempered before being placed into service. The harder forms of tempered martensite have poorer machinability.

NITRIDING

A heat treating method in which nitrogen is diffused into the surface of iron-base alloys. This is done by heating the metal at a temperature of about 950°F in contact with ammonia gas or other suitable nitrogenous materials. The surface, because of formation of nitrides, becomes much harder than the interior. Depth of the nitrided surface is a function of the length of time of exposure and can vary from .0005" to .032" thick. Hardness is generally in the 65 to 70R range, and therefore these structures are almost always ground.

NITRIDING STEELS

Steels which are selected because they form good case hardened structures in the nitriding process. In these steels, elements such as aluminum and chromium are important for producing a good

NODULAR IRONS

See Ductile Irons.

NORMALIZING

Heat treatment of iron-base alloys above the critical temperature, followed by cooling in still air. This is often done to refine or homogenize the grain structure of castings, forgings, and wrought steel products.

0

A letter listed along with the composition of aluminum and magnesium alloys indicating that the alloy has been annealed.

OHFC

A designation applied to indicate oxygen free, high conductivity copper.

PEARLITE

A microconstituent found in iron-base alloys consisting of a lamellar (platelike) composite of ferrite and iron carbide. This structure results from the decomposition of austenite and is very common in cast irons and annealed steels.

PLAIN CARBON STEELS

Steels with carbon plus small amounts of manganese, silicon, sulfur, phosphorus, or other elements.

PRECIPITATION Certa
HARDENING spond
STAINLESS STEELS ment.

Certain stainless steel compositions which respond to precipitation hardening or aging treatment

PRESSED

The "as pressed" condition of alloys when compacted from powders. Pressing of powdered metals is generally followed by sintering. Some difficult or impossible to machine materials such as tungsten and carbides are shaped by various machining operations in the "green" condition, generally produced by low temperature presintering.

QUENCHED

Alloys which are rapidly cooled by immersion in liquids or gases after heating.

REFRACTORY ALLOYS Alloys or elements which have melting temperatures above 4000°F. The most common refractory alloys of current industrial interest are tungsten, molybdenum, columbium, and tantalum.

RESULFURIZED ALLOYS

Alloys which have sulfur added to improve machinability.

ROCKWELL HARDNESS

Rockwell hardness is a measure of the difference in depth of penetration of an indenter between a major and a minor load. Normally a minor load of 10 kilograms is applied, followed by the application and release of a major load which will vary from 60 to 150 kilograms. A direct reading dial indicates the increase in penetration caused by the application of the major load. This dial has various scales for use with the various major loads and penetrators. The more commonly used Rockwell scales are Rockwell "C" (R_c), using a diamond Brale penetrator and a 150 kilogram major load, and Rockwell "B" (RB), using a 1/16" diameter ball penetrator and a major load of 100 kilograms. Other scales used in this data book are R_A , R_E , R_R , and R_M .

SINTERED

The sintered condition results from a heating of pressed powdered metals for specified times at elevated temperature. Improved strength and other benefits result, but generally machinability is decreased.

SOLUTION TREATED

A process in which it is possible to dissolve microconstituents by taking certain alloys to an elevated temperature and then keeping them in solution after quenching. Often a solution treatment is followed by a precipitation or aging treatment to improve the mechanical properties. Most high temperature alloys which are solution treated and aged machine better in the solution treated state just before they are aged.

STAINLESS STEELS Steels which have good or excellent corrosion resistance. One of the common grades contains 18% chromium and 8% nickel. There are three broad classes of stainless steels -- ferritic, austenitic, and martensitic. These various classes are produced through the use of various alloying elements in differing quantities.

STRESS RELIEVED

The heat treatment used to relieve the internal stresses induced by forming or heat treating operations. It consists of heating a part uniformly, followed by cooling slow enough so as not to reintroduce stresses. To obtain low stress levels in steels and cast irons, temperatures as high as 1250°F may be required.

T

A letter listed along with the composition of aluminum and magnesium alloys indicating that the alloy has been subjected to heat treatment. T4, T5, T6, etc. indicate various specific heat treatments.

TH

Letters used with the composition of precipitation hardening stainless steels designating special heat treatments using in part a refrigeration cycle to develop specific properties.

TEMPERED

Reheating of hardened steels or hardened cast irons to reduce hardness and to lower internal stress. The temperature used depends upon the mechanical properties specified but generally ranges from 300 to 1200°F.

THERMOPLASTIC

A plastic which can be repeatedly softened by an increase in temperature and hardened by a decrease in temperature. Heating causes physical changes rather than chemical.

THERMOSETTING PLASTIC

A plastic which upon heating is changed chemically into a material having new properties.

TITANIUM ALLOYS

Alloys of the element titanium. They combine light weight, excellent strength, moderately good high temperature strength, and excellent corrosion resistance. Various alloying elements and heat treatments provide differing crystal structures. Depending upon the crystal structure of the finished alloy, they are classified as alpha, beta, or alphabeta alloys.

TITANIUM CAR BIDE Carbides in which the greatest amount of carbide is titanium instead of tungsten and which also contain other carbides plus nickel and molybdenum as binders.

TOOL STEELS

Steels used to make tools of all types and dies. Many of these steels have considerable quantities of alloying elements such as chromium, carbon, tungsten, molybdenum, and other elements. These form hard carbides which provide good wearing qualities but at the same time decrease machinability. Tool steels in the trade are classified for the most part by their applications, such as hot work die, cold work die, high speed, shock resisting, and mold steels.

ULTIMATE STRENGTH

The maximum stress, expressed in pounds per square inch, which a material will carry before breaking under a slowly applied, continually increasing load.

ULTRA HIGH

Those steels intended for designs requiring great STRENGTH STEELS strength (generally greater than 200, 000 psi) such as landing gears, rocket motor cases, etc. This class includes some of the alloy steels and tool steels heat treated to high hardness as well as the hardened maraging steels.

WROUGHT

To indicate alloys which have been mechanically worked in their manufacture in contrast with the cast condition.

YIELD STRENGTH

Yield strength is the stress obtained in a tensile or compressive test of materials. At this stress, materials begin to take a permanent set, i. e. they will not return to their original shape after a load is removed.

DESCRIPTION OF AFMDC

AIR FORCE MACHINABILITY DATA CENTER, 3980 Rosslyn Drive, Cincinnati, Onio 45209. Operated for the Air Force Materials Laboratory, Manufacturing Technology Division, under Contract AF 33(815)-5282, by Metcut Research Associates Inc.

SCOPE

The Air Force Machinability Data Center (AFMDC) collects, evaluates, stores, and disseminates material removal information including specific and detailed machining data for the benefit of industry and government. Strong emphasis is given to engineering evaluation for the purpose of developing optimized material removal parameters, such as speeds, feeds, depths of cut, tool material and geometry, cutting fluids and other significant variables. Data are being processed for all types of materials and for all kinds of material removal operations such as turning, milling, drilling, tapping, grinding, electrical discharge machining, electrochemical machining, etc.

COLLECTION

AFMDC has a mechanized system in which punch cards are used to store and retrieve all types of material removal information including all significant numerical data. An IBM 1130 computing system is being used for storing and processing data from a master card and disk file and for computer decoding. The focal concept for acquisition, interrogation, or presentation of information is the specific material (with definite chemical, physical, or mechanical properties) and the specific material removal operation being used. When necessary, card source control codes may be used to retrieve original documents which are in document storage at AFMDC.

INFORMATION SERVICES

AFMDC places strong emphasis on providing specific and detailed answers to technical inquiries in the field of material removal. A User File, consisting of important users in the field of material removal, has been developed to receive information products including machining data pamphlets and tables on materials of current interest, state-of-the-art reports, technical announcements, and other appropriate items. Services are available without charge to the aerospace industry, Department of Defense (including all of the military services and their contractors), and other government agencies, technical institutions, and non-military industries in a position to assist the defense effort.

TO REQUEST MACHINING INFORMATION

TELEPHONE:

513-271-9510

TWX:

810-461-2840 or

WRITE:

Air Force Machinability Data Center

3980 Rosslyn Drive Cincinnati, Ohio 45209

TO HELP US ANSWER YOUR INQUIRY, IF POSSIBLE PLEASE:

- 1. Identify the material being machined (specification or trade name); condition (as cast, hot rolled, cold drawn, annealed, quenched and tempered, etc.); microstructure and hardness.
- 2. Identify the material removal operation in question (turning, milling, drilling, tapping, surface grinding, electrical discharge machining (EDM), electrochemical machining (ECM), etc.).
- Specify reasons for requiring data unless your needs are proprietary. This enables AFMDC to broaden the scope of its technical advice.
- 4. Specify delivery requirements.
- 5. Indicate to whom the inquiry reply should be sent.
- Transmit all details concerning present practices, including feeds, speeds, cutting tool
 material and geometry, cutting fluids, etc., in the event your inquiry pertains to
 improvement of an existing machining situation.

NOTE: Association of the names of companies and individuals with specific requests is kept confidential. However, data developed remain the property of AFMDC for dissemination as required for answering similar inquiries and for developing data products.

OTHER SOURCES OF MACHINING DATA*

- 1. AIR FORCE MACHINABILITY DATA CENTER. See Page A-17 for a complete description of services available.
- 2. MACHINING DATA HANDBOOK, Metcut Research Associates Inc. Cincinnati: 1966.
- 3. MACHINERY'S HANDBOOK, by E. Oberg and F.D. Jones. 17th ed. New York: Industrial Press, 1964.
- 4. MACHINING DIFFICULT ALLOYS, American Society for Metals. Metals Park, Ohio: The Society, 1982.
- 5. MACHINING THE SPACE-AGE METALS, American Society of Tool and Manufacturing Engineers. Dearborn, Michigan: The Society, 1985.
- 6. MACHINING WITH CARBIDES AND OXIDES, American Society of Tool and Manufacturing Engineers. New York: McGraw-Hill, 1962.
- 7. METAL CUTTING BIBLIOGRAPHY, 1943-1958, John Crerar Library, Chicago. Research Information Service. Detroit: American Society of Tool and Manufacturing Engineers, 1960.
- 8. METAL CUTTING TOOL HANDBOOK, Metal Cutting Tool Institute. New York: The Institute. 1965.
- 9. NEW AMERICAN MACHINIST'S HANDBOOK, Rupert LeGrand. New York: McGraw-Hill, 1955.
- 10. NUMERICAL CONTROL IN MANUFACTURING, American Society of Tool and Manufacturing Engineers. New York: McGraw-Hill, 1963.
- 11. TOOL ENGINEERS HANDBOOK, American Society of Tool and Manufacturing Engineers.
 2d ed. New York: McGraw-Hill, 1959.
- 12. TOOLING FOR AIRCRAFT AND MISSILE MANUFACTURE, American Society of Tool and Manufacturing Engineers. New York: McGraw-Hill, 1984.

Partial list of important sources in addition to those listed on Page vi from which machining data were extracted for this report.

Unclassified

Seci		CI	:	C:	4:
26.01	ITIIV		3881	HCA	HOD

Security Classification		
	NTROL DATA - R&D	arterior male an area and an
(Security classification of title, body of abstract and indexi 1. ORIGINATING ACTIVITY (Corporate author)	التكاف التسميد المتهاب المثر والتنفيذ والمستقيلة فالمتعار أفو الإسراء المستهدي ويستهد والمتعار	the overall report is classified) RT SECURITY CLASSIFICATION
Air Force Machinability Data Center		iclassified
Metcut Research Associates Inc.	2 b GROU	
Cincinnati, Ohio 45209	N/	
3 REPORT TITLE		
Machining Data for Numerical Control	- Reaming	
4. DESCRIPTIVE NOTES (Type of report and inclusive dates) Tool Life test data - reaming of acros	pace materials	
5. AUTHOR(S) (Lest name, lirst name, initial) Field, Michael Mehl, Clarence L. Kahles, John F.		
6. REPORT DATE	74. TOTAL NO. OF PAGES	76. NO. OF REFS
November 1966	36	6
8a. CONTRACT OR GRANT NO.	94. ORIGINATOR'S REPORT NUM	ABER(S)
AF 33(615)-5262 b. project No.	AFMDC 66-1	1.7
9-700		
c.	96. OTHER REPORT NO(S) (Any this report)	other numbers that may be essigned
d.		
10. AVAILABILITY/LIMITATION NOTICES Qualified r	equestors may obtain	copies of this report
from DDC. This document is subject foreign governments or foreign nations of the Air Force Materials Laboratory	to export-controls and als may be made only	d each transmittal to
11. SUPPL EMENTARY NOTES	12. SPONSORING MILITARY ACT	
	Manufacturing Tec	0,
	Air Force Materia Wright-Patterson	Air Force Base, Ohio
	Wilght-Fatterson	All Force Dase, Onto
This report presents an extensive set	of machining data sal	acted from six
USAF Machinability Reports. Data ar		_
including machining variables such as	_	· · · · · · · · ·
fluid, depth, feed, and tool life end po		· ·
relationship between tool life and cutti		
sets of data, thereby making it possibl	-	-
or minimum cost. While these data as	•	
assistance in providing data for numer		•
of great value in any type of machining	situation involving th	ne materials for
which machining data are presented. Series, pertains to reaming.	Specifically this repor	rt, the last of a
i		
The previous reports in this series we Milling, Drilling, Peripheral End Mill (Report Nos. AFMDC 66-1.1 through 6 a single volume (Report No. AFMDC 6	ing, End Mill Slotting 6-1.6). All reports	g, and Tapping

DD . FORM 1473

Unclassified

Security Classification

Unclassified

A STATE OF THE PROPERTY OF THE

MEN WORDS	LIN	KA	LINE	(B	LIN	КС
KEY WORDS	ROLE	₩T	ROLE	wT	ROLL	WT
Reaming						
Numerical Control						
Machining Data						
Tool Life Data]			
Alloy Steels					}	
Stainless Steels		l.	1 1			
Titanium Alloys						ł
High Temperature Alloys			1			
Refractory Alloys						
Refrasil			1		1	
			1			
					1 1	

INSTRUCTIONS

- 1. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (corporate author) issuing the report.
- 2a. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.
- 2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.
- 3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.
- 4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.
- 5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.
- 6. REPORT DATE: Enter the date of the report as day, month, year, or month, year. If more than one date appears on the report, use date of publication.
- 7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.
- 7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.
- 8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.
- 8b, 8c, & 8d. PROJECT NUMBER: Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.
- 9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.
- 9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (either by the originator or by the sponsor), also enter this number(s).
- 10. AVAILABILITY/LIMITATION NOTICES: Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known

- 11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.
- 12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (paying for) the research and development. Include address.
- 13. ABSTRACT: Enter an abstract giving a brief and factual aummary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rules, and weights is optional.

Unclassified

Security Classification